| QPS requirements |   |  |   |  |                    |   | Information |   |
|------------------|---|--|---|--|--------------------|---|-------------|---|
| Test             |   | Tolerance(s)   | Flight condition  | Test details   | Simulator<br>level |   | tor         | Notes   |
| Entry No.        | Title   |  | -   |  | в                  | С | D           |   |
| 2.a.1            | Cyclic  | Breakout—±0.25 lbs. (0.112<br>daN) or 25%; Force—±1.0<br>lb. (0.224 daN) or 10%. | Ground; Static conditions with<br>the hydraulic system (if ap-<br>plicable) pressurized; sup-<br>plemental hydraulic pres-<br>surization system may be<br>used. Trim On and Off.<br>Friction Off Augmentation<br>(if applicable) On and Off.  | Record results for an uninter-<br>rupted control sweep to the<br>stops. (This test does not<br>apply if aircraft hardware<br>modular controllers are<br>used.) | x                  | × | x           | Flight Test Data for this test<br>does not require the rotor<br>to be engaged/turning. The<br>phrase "if applicable" re-<br>garding stability augmenta-<br>tion systems means if an<br>augmentation system is<br>available and if this system<br>may be operational on the<br>ground under static condi-<br>tions as described here.              |
| 2.a.2            | Collective/Pedals                               | Breakout—±0.5 lb. (0.224<br>daN) or 25%; Force—±1.0<br>lb. (0.224 daN) or 10%.   | Ground; Static conditions with<br>the hydraulic system (if ap-<br>plicable) pressurized; sup-<br>plemental hydraulic pres-<br>surization system may be<br>used. Trim On and Off.<br>Friction Off. Augmentation<br>(if applicable) On and Off. | Record results for an uninter-<br>rupted control sweep to the<br>stops.  | x                  | x | x           | Flight Test Data for this test<br>does not require the rotor<br>to be engaged/turning. The<br>phrase "if applicable" re-<br>garding stability augmenta-<br>tion system means if a sta-<br>bility augmentation system<br>is available and if this sys-<br>tem may be operational on<br>the ground under static<br>conditions as described<br>here. |
| 2.a.3            | Brake Pedal Force vs. Posi-<br>tion.            | ±5 lbs. (2.224 daN) or 10%.  | Ground; Static conditions.  |  | х                  | х | х           |   |
| 2.a.4            | Trim System Rate (all appli-<br>cable systems). | Rate—±10%.   | Ground; Static conditions.<br>Trim On, Friction Off.  | The tolerance applies to the recorded value of the trim rate.  | х                  | х | x           |   |

## TABLE C2A—FULL FLIGHT SIMULATOR (FFS) OBJECTIVE TESTS—Continued

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