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2.a.2	Collective and Pedals	Breakout—±0.5 lb (0.224 daN) or 25%. Force—±1.0 lb (0.224 daN) or 10%.	Ground; Static conditions. Trim On and Off. Friction Off. Augmentation On and Off.	Record results for an uninter- rupted control sweep to the stops.	X	Х	Х	
2.a.3	Brake Pedal Force vs. Position.	±5 lbs (2.224 daN) or 10%	Ground; Static conditions.		х	х	х	
2.a.4	Trim System Rate (all applicable systems).	Rate—±10%	Ground; Static conditions. Trim On. Friction Off.	The tolerance applies to the recorded value of the trim rate.	x	х	х	
2.a.5	Control Dynamics (all axes).	$\pm 10\%$ of time for first zero crossing and ± 10 (N + 1)% of period thereafter. $\pm 10\%$ of amplitude of first overshoot. $\pm 20\%$ of amplitude of 2nd and subsequent overshoots greater than 5% of initial displacement. ± 1 overshoot.	Hover/Cruise Trim On Friction Off.	Results must be recorded for a normal control displacement in both directions in each axis, using 25% to 50% of full throw.		x	X	Control Dynamics for versible control sys may be evaluated ground/static condi Refer to paragraph this attachment for tional information. It the sequential peri- full cycle of oscillat
2.a.6	Freeplay	±0.10 in. (±2.5 mm)	Ground; Static conditions.	Record and compare results for all controls.	х	х	х	
2.b	Low Airspeed Handling Qualities.							
2.b.1	Trimmed Flight Control Positions.	Torque ±3% Pitch Attitude ±1.5° Bank Attitude ±2° Longitudinal Control Position ±5% Lateral Control Position ±5% Direc- tional Control Position ±5% Collective Control Position ±5%.	Translational Flight IGE—Sideward, rearward, and for- ward flight. Aug- mentation On and Off.	Record results for several air- speed increments to the translational airspeed limits and for 45 kts. forward airspeed. May be a series of snapshot tests.			x	
2.b.2	Critical Azimuth	Torque ±3% Pitch Attitude ±1.5°, Bank Attitude ±2°, Longitudinal Control Position ±5%, Lateral Control Position ±5%, Direc- tional Control Position ±5%, Collective Control Position ±5%.	Stationary Hover. Augmentation On and Off.	Record results for three relative wind directions (including the most critical case) in the critical quadrant. May be a series of snapshot tests.			x	
2.b.3	Control Response.	•		•				