TABLE D2A—FLIGHT TRAINING DEVICE (FTD) OBJECTIVE TESTS—Continued

			QPS requirements					
Test	- Tolerances	Flight conditions	Test details	FTD level		vel	Notes	
Title				5	6	7	ivoles	
Visual System								
			Visual System Response Time Test.					
Latency.								
	150 ms (or less) after helicopter response.	Takeoff, climb, and descent.	One test is required in each axis (pitch, roll and yaw) for each of the three conditions (take-off, cruise, and approach or landing).			х		
Transport Delay.								
	150 ms (or less) after controller movement.	N/A	A separate test is required in each axis (pitch, roll, and yaw).			Х		
Field-of-view.								
Reserved.								
Continuous visual field- of-view.	Minimum continuous field-of-view providing 146° horizontal and 36° vertical field-of-view for each pilot simultaneously and any geometric error between the Image Generator eye point and the pilot eye point is 8° or less.	N/A	An SOC is required and must explain the geometry of the installation. Horizontal field-of-view must not be less than a total of 146° (including not less than 73° measured either side of the center of the design eye point). Additional horizontal field-of-view capability may be added at the sponsor's discretion provided the minimum field-of-view is retained. Vertical field-of-view: Not less than a total of 36° measured from the pilot's and co-pilot's eye point.			×	Horizontal field-of-view is centered on the zero de- gree azimuth line relative to the aircraft fuselage.	
-	Visual System Visual System Response This test is also sufficient Latency. Fransport Delay. Field-of-view. Reserved. Continuous visual field-	Title //isual System Response Time: (Choose either test 4.a.1. or 4 This test is also sufficient for flight deck instrument response ti atency. 150 ms (or less) after helicopter response. 150 ms (or less) after controller movement. Field-of-view. Reserved. Continuous visual field-of-view. Minimum continuous field-of-view providing 146° horizontal and 36° vertical field-of-view for each pilot simultaneously and any geometric error between the Image Generator eye point is 8° or end of the pilot eye pilot end of the pilot eye point is 8° or end of the pilot eye pilot end of	Title //isual System //isual System Response Time: (Choose either test 4.a.1. or 4.a.2. to satisfy test 4.a., This test is also sufficient for flight deck instrument response timing.) atency. 150 ms (or less) after helicopter response. Takeoff, climb, and descent. 150 ms (or less) after controller movement. N/A	Title Visual System Visual System Response Time: (Choose either test 4.a.1. or 4.a.2. to satisfy test 4.a., Visual System Response Time Test. This test is also sufficient for flight deck instrument response timing.)	Title Visual System Visual System Response Time: (Choose either test 4.a.1. or 4.a.2. to satisfy test 4.a., Visual System Response Time Test. This test is also sufficient for flight deck instrument response timing.) atency. 150 ms (or less) after helicopter response. 150 ms (or less) after controller movement. 150 ms (or less) after controller movement. 150 ms (or less) after controller movement. A separate test is required in each axis (pitch, roll and yaw) for each of the three conditions (take-off, cruise, and approach or landing). Field-of-view. Reserved. Continuous visual field-of-view providing 146° horizontal and 36° vertical field-of-view for each pilot simultaneously and any geometric error between the Image Generator eye point and the pilot eye point is 8° or less. An SOC is required and must explain the geometry of the installation. Horizontal field-of-view must not be less than a total of 146° (including not less than 173° measured either side of the center of the design eye point). Additional horizontal field-of-view in retained. Vertical field-of-view is retained. Vertical field-of-view measured from the pilot's	Title	Title	