

Table A2A - Full Flight Simulator (FFS) Objective Tests									
QPS REQUIREMENTS									
Test		Tolerance	Flight Conditions	Test Details	Simulator Level				Notes
Entry Number	Title				A	B	C	D	
		reversible flight control systems; ±2.2 daN (5 lbf) or ±10% of column force; ±3 kt airspeed; ±1.5° pitch angle; ±1.5° AOA; ±6 m (20 ft) height; ±2° roll angle; ±2° side-slip angle; ±3° heading angle; For airplanes with reversible flight control systems: ±2.2 daN (5 lbf) or ±10% of column force; ±1.3 daN (3 lbf) or ±10% of wheel force; and ±2.2 daN (5 lbf) or ±10% of rudder pedal force.							
1.b.5	Critical engine failure on take-off.		Takeoff.	Record takeoff profile to at least 61 m (200 ft) AGL. Engine failure speed must be within ±3 kt of airplane data. Test at near maximum takeoff weight.	X	X	X	X	
1.b.6	Crosswind takeoff.	±3 kt airspeed; ±1.5° pitch angle; ±1.5° AOA; ±6 m (20 ft) height; ±2° roll angle.	Takeoff.	Record takeoff profile from brake release to at least 61 m (200 ft) AGL. This test requires test data, including wind profile, for a crosswind component of at least 60% of the airplane performance data value measured at 10 m (33 ft) above the runway. Wind components must be provided as headwind	X	X	X	X	In those situations where a maximum crosswind or a maximum demonstrated crosswind is not known, contact the NSPM.