			0 "	5 1 4
Parameters	Range	Installed system 1 minimum accuracy (to recovered data)	Sampling inter- val (per second)	Resolution 4 read out
Roll attitude	±60° or 100% of usable range, whichever is greater.	±2°	1	0.8°
Stabilizer trim position	Full range	±3% unless higher uniquely required.	1	1%³.
Or Pitch control position	Full range	±3% unless higher uniquely required.	1	1% 3.
Engine Power, Each Engine				
Fan or N <sub>1</sub> speed or EPR or cockpit indications used for aircraft certification. Or	Maximum range	±5%	1	1% <sup>3</sup> .
Prop. speed and torque (sample once/sec as close together as practicable).			1 (prop speed), 1 (torque).	
Altitude rate 2 (need depends on altitude resolution).	±8,000 fpm	±10%. Resolution 250 fpm below 12,000 ft. indicated.	1	250 fpm Below 12,000
Angle of attack 2 (need depends on altitude resolution).	-20° to 40° or of usable range.	±2°	1	0.8%3
Radio transmitter keying (discrete).	On/off		1.	
TE flaps (discrete or analog)	Each discrete position (U, D, T/O, AAP). Or.		1.	
	Analog 0-100% range	±3°	1	1%3
LE flaps (discrete or analog)	Each discrete position (U, D, T/O, AAP). Or.		1.	
	Analog 0-100% range	±3°	1	1% <sup>3</sup> .
Thrust reverser, each engine (Discrete).	Stowed or full reverse		1.	
Spoiler/speedbrake (discrete)	Stowed or out		1.	
Autopilot engaged (discrete)	Engaged or disengaged		1.	

<sup>&</sup>lt;sup>1</sup>When data sources are aircraft instruments (except altimeters) of acceptable quality to fly the aircraft the recording system excluding these sensors (but including all other characteristics of the recording system) shall contribute no more than half of the values in this column.

<sup>2</sup>If data from the altitude encoding altimeter (100 ft. resolution) is used, then either one of these parameters should also be recorded. If however, altitude is recorded at a minimum resolution of 25 feet, then these two parameters can be omitted.

<sup>3</sup>Per cent of full range.

<sup>4</sup>This column applies to aircraft manufacturing after October 11, 1991.

 $[\mathrm{Doc.\ No.\ 25530,\ 53\ FR\ 26152,\ July\ 11,\ 1988;\ 53\ FR\ 30906,\ Aug.\ 16,\ 1988,\ as\ amended\ by\ Amdt.\ 13569,\ 62\ FR\ 38397,\ July\ 17,\ 1997]}$ 

APPENDIX C TO PART 135—HELICOPTER FLIGHT RECORDER SPECIFICATIONS

Parameters	Range	Installed system 1 minimum accuracy (to recovered data)	Sampling interval (per second)	Resolution <sup>3</sup> read out
Relative time (from recorded on prior to takeoff).	25 hr minimum	±0.125% per hour	1	1 sec.
Indicated airspeed	$V_{\rm m}$ in to $V_{\rm D}$ (KIAS) (minimum airspeed signal attainable with installed pilot-static system).	±5% or ±10 kts., whichever is greater.	1	1 kt.
Altitude	-1,000 ft. to 20,000 ft. pressure altitude.	±100 to ±700 ft. (see Table 1, TSO C51-a).	1	25 to 150 ft.
Magnetic heading	360°	±5°	1	1°.
Vertical acceleration	-3g to + 6g	±0.2g in addition to ±0.3g maximum datum.	4 (or 1 per sec- ond where peaks, ref. to 1g are re- corded).	0.05g.
Longitudinal acceleration	±1.0g	±1.5% max. range excluding datum error of ±5%.	2	0.03g.
Pitch attitude	100% of usable range	±2°	1	0.8°.
Roll attitude	±60° or 100% of usable range, whichever is greater.	±2°	1	0.8°.